

FROM SEPARATE STREAM TURBOFAN ANALYSIS:

$$\frac{F}{\dot{m}_c + \dot{m}_F} = \frac{980.3}{1 + 8.36} \left\{ \left[ \frac{2}{\gamma-1} \frac{7}{1.968(2.17)} (1.968(2.17)(0.399) - 1) \right]^{\frac{1}{2}} - 2.2 \right. \\ \left. + 8.36 \left( \left[ \frac{2}{\gamma-1} (1.968(1.12) - 1) \right]^{\frac{1}{2}} - 2.2 \right) \right\}$$

$$\frac{F}{\dot{m}_c + \dot{m}_F} = 235.6 \frac{\text{lb} \cdot \text{s}}{\text{slug}} = \underline{\underline{7.32}} \frac{\text{lb} \cdot \text{s}}{\text{lbm}}$$

$$f = \frac{(6006) \cdot 400}{4.89(10)^8} (7 - 1.968(2.17)) = 0.0134$$

$$S^* = \frac{3600(0.0134)}{(1 + 8.36)(7.32)} = \underline{\underline{0.704}} \frac{\text{lbm}}{\text{lb} \cdot \text{hr}}$$

c) RECALCULATE w/  $\gamma_{\lambda b} = 6.8$ .

THE FAN STREAM WILL NOT BE AFFECTED BY THE AFTERBURNING.

FOR THE CORE STREAM, FOR NO AFTERBURNING:

$$\left. \frac{U_g}{U_0} \right)_0 = \frac{1}{2.2} \left[ \frac{2}{\gamma-1} \frac{7}{1.968(2.17)} (0.399(1.968)(2.17) - 1) \right]^{\frac{1}{2}}$$

$$\left. \frac{U_g}{U_0} \right)_0 = 1.058$$

FROM THE EQUATION DERIVED IN (a):

$$\frac{U_g/U_0}{(U_g/U_0)_0} = \left[ \frac{6.8}{7 - 1.968(2.17) - 1} \right]^{\frac{1}{2}} = 1.203$$

$$U_g/U_0 = 1.203(1.058) = 1.273$$

FOR THE FAN STREAM:

$$\frac{U_g'}{U_0} = \left[ \frac{1}{M_0^2} \frac{2}{\gamma-1} (\gamma_c \gamma_r - 1) \right]^{\frac{1}{2}} \\ = \frac{1}{2.2} \left[ \frac{2}{\gamma-1} (1.968(1.12) - 1) \right]^{\frac{1}{2}} = 1.115$$

$$\frac{F}{\dot{m}_c + \dot{m}_F} = \frac{a_0 M_0}{1 + \alpha} \left[ \left( \frac{U_g}{U_0} - 1 \right) + \alpha \left( \frac{U_g'}{U_0} - 1 \right) \right]$$

$$\frac{F}{\dot{m}_c + \dot{m}_F} = \frac{(980.3)(2.2)}{1 + 8.36} \left[ (1.273 - 1) + 8.36(1.115 - 1) \right]$$

$$\frac{F}{\dot{m}_c + \dot{m}_F} = \underline{\underline{8.84}} \frac{\text{lb} \cdot \text{s}}{\text{lbm}}$$